Quantification of Grid Refinement Effects for NASA High Lift Trap Wing Using Error Transport Model

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Presentation Overview

- Analysis Objectives
 - Assess refinement effects as we approach stall
 - How well do our error prediction methods work?
 - How can we address/identify dissimilar refinement pairs?
- Error Quantification Method ETE Solver
- Discussion of FUN3D results
 - 28 through 36 degrees angle of attack
 - Coarse/Medium/Fine grids
- Error Predictions Using ETE
 - FUN3D Grid Sequence
 - Cell-centered ETE on *USM3D* Solution
- New Approach for Assessment of Refinement Pairs
- Conclusions and Lessons Learned



Errors in CFD Analysis

- Solution errors inherent to any CFD analysis
 - Discretization errors (grid size/spacings, time step)
 - Modeling errors (turbulence, transition, *etc*.)
 - Usage, iterative convergence, coding errors, *etc*.
- Discretization errors generally most dominant, but must be understood before tackling turbulence modeling issues
- How do we check our results? Verification & Validation (V&V)
 - Compare with test data (validation)
 - Perform grid refinement study (verification) required workshop element!
 - Richardson extrapolation given 3 mesh sequence
- Our prior work at HLPW-1 examined refinement sequences at 13°, 28°
 - Error transport model seemed capable of predicting increments between coarse/fine and medium/fine solutions for these conditions
 - Current follow-on explores increments near stall, maximum C_L



Study Objectives

- Explore mesh dependent effects as we approach stall
 - UT5 tetrahedral grid sequence employed
 - FUN3D solver, Spalart-Allmaras turbulence model
- Apply Error Transport Model to solution sequences
 - Evaluate method's ability to predict increments between solutions
 - If grid-induced errors can be predicted reliably, it potentially precludes need to run fine grid solution
 - In addition, reliable prediction could confirm confidence in results
- Investigate how to quantify dissimilar solution pairs
 - Such pairs cannot be considered in refinement sequence
 - ETE method cannot account for such disparities
 - Potential approach developed with preliminary results shown
- Identify shortcomings that remain to be addressed



Error Prediction / Quantification

• 3D Error Transport Equation (ETE) Solver for steady state flows

$$\frac{\partial}{\partial t} \iiint \left(\vec{Q} - \vec{Q}^h \right) dV + \iint \left(\left(\vec{F} \left(\vec{Q} \right) - \vec{F} \left(\vec{Q}^h \right) \right) \cdot \hat{n} \right) dA = \iint \left(\left(\vec{G} \left(\vec{Q} \right) - \vec{G} \left(\vec{Q}^h \right) \right) \cdot \hat{n} \right) dA - \vec{R}$$

 $\frac{\partial}{\partial t} \iiint \vec{\varepsilon} \, dV + \iiint \left(\left(A \left(\vec{Q}^h \right) \vec{\varepsilon} \right) \cdot \hat{n} \right) dA = \iint \left(\vec{G} \left(\vec{\varepsilon} \right) \cdot \hat{n} \right) dA - \vec{R}_{INV} - \vec{R}_{TURB}$

- Inviscid residual: upwind terms of Roe flux
- Turbulent residual: accounts for effects of error in μ_t on mean flow
- k– ϵ , k– ω , Spalart-Allmaras models supported
- Recently expanded to support cell-centered solvers and solve ETE using cell-centered or node-centered discretization^{*}
- Error Function Library
 - Propagates predicted errors into derived variables of interest
 - PLOT3D functions, integrated functions, etc.

^{*} Cavallo, P.A., O'Gara, M.R., Feldman, G.M., and Liu, Z., "Unified Error Transport Equation Solver for Solution Verification on Unstructured Grids," AIAA Paper 2012-3345, 42nd Fluid Dynamics Conference, New Orleans, LA, June 25-28, 2012.



What Are We Looking For?

- Goal of Error Transport research is to establish alternative solution verification method
 - Provide reliable predictions of mesh-induced errors
 - Prediction of coarse-to-fine grid increments
 - Useful for quantifying local and integrated quantities
- Error bars predicted by ETE solution and Error Functions should:
 - 1) Contain fine grid results
 - 2) Contain results of Richardson extrapolation
 - 3) Decrease in magnitude with grid refinement
 - 4) Not be overly conservative as to be unusable
- If fine grid results fall outside predicted error bars, it potentially indicates new flow features result from grid refinement
- If test data falls outside predicted solution and error bars, it potentially indicates a deficiency in physical modeling



Lift Characteristics Near Stall

- Maximum C_L predicted to occur at 32 degrees for each grid
- Stall point is mesh dependent
 - Medium grid stalls first just beyond 34 degrees
 - Coarse and fine grids both stall at ~35 degrees
 - Separation patterns on coarse, fine grids are quite different





FUN3D Results, $\alpha = 28^{\circ}$





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FUN3D Results, α =32°





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FUN3D Results, α =34°





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FUN3D Results, α =35°





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FUN3D Results, α =36°





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Stall Patterns





ETE Results, $\alpha = 28^{\circ}$





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ETE Results, $\alpha = 32^{\circ}$





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ETE Results, $\alpha = 34^{\circ}$





ETE Results, $\alpha = 35^{\circ}$





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ETE Results, $\alpha = 36^{\circ}$





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USM3D ETE Results, α =28°





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- Error Transport model cannot account for absence or presence of a flow structure due to grid refinement
 - e.g., stall vs. no stall
- Means of quantifying similarly between solution pairs needed to assess if sequence is near monotonic range – otherwise Richardson extrapolation and ETE are not applicable
 - e.g., USM3D solution sequences beyond
 30 degrees (Pandya et al., 2011)
- Approach explored based on intersection of metric ellipsoids
 - Metric formed using Hessian matrix of 2nd derivatives
 - Eigenanalysis extracts principal directions and length scales associated with flowfield





- Metric tensors represented as 3D ellipsoid
- Intersection of ellipsoids can indicate scaling and alignment
 - Consider M1 to be coarse solution metric, M2 fine solution metric
 - Solution should sharpen with mesh refinement: principal directions are nearly aligned and length scales are merely scaled from coarse to fine grid
 - Misalignment between solution pairs would result in intersected volume that is less than volume of metric ellipsoid M2



• Examine the volume fraction $\phi = V_{INT} / V_2$



Similar Solution Pair, α =28°

- Preliminary results comparing Coarse (top) and Medium (bottom) grid solutions from USM3D
- Solution pair is similar and volume fraction is consistent with this
- "Spottiness" of plot believed related to data transfer/interpolation







Dissimilar Solution Pair, α =32°

- At 32 degrees, the Coarse USM3D solution predicts stall, while the Medium grid predicts attached flow
- Volume fraction from metric intersection comparing the solutions picks up this disparity well
- Still exploring how we can improve method and use this information





Conclusions and Lessons Learned

- Investigated grid refinement effects up to and beyond maximum C_L
 - Angle of peak C_L itself is grid-independent (~32 degrees)
 - Onset of stall, separated flow structures exhibit mesh dependent behavior
 - Inboard vs. outboard evolution of separation
- High alpha cases were a considerable challenge for Error Transport Model's ability to capture grid-induced increments
 - Generally, predicted errors in C_p were quite large for all medium grid FUN3D solutions
 - Clearly, accuracy concerns must be revisited for these cases
 - Cell-centered ETE solution for USM3D shows promise
- Path towards quantifying solution similarity established
 - Method based on computing metric tensor intersections
 - Preliminary application on USM3D sequence
 - Method needs further development and testing



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- Questions?
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